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[Rules and Regulations] [Page 17905-17906]

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#### DEPARTMENT OF TRANSPORTATION

**Federal Aviation Administration** 

**14 CFR Part 39** 

[Docket No. 2002-NM-160-AD; Amendment 39-13560; AD 2004-07-16]

**RIN 2120-AA64** 

Airworthiness Directives; Construcciones Aeronauticas, S.A. (CASA), Model C-235 Series Airplanes

**AGENCY:** Federal Aviation Administration, DOT.

**ACTION:** Final rule.

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**SUMMARY:** This amendment adopts a new airworthiness directive (AD), applicable to certain CASA Model C-235 series airplanes, that requires modification of the electrical wiring of the rudder trim control unit. This action is necessary to prevent the flight crew from being able to inhibit the aural warning for the landing gear up. If the flight crew of the next flight or possibly of the same flight is unaware that the aural warning had been disabled, they could inadvertently land the airplane with the landing gear not down and locked. This action is intended to address the identified unsafe condition.

**DATES:** Effective May 11, 2004.

The incorporation by reference of certain publications listed in the regulations is approved by the Director of the Federal Register as of May 11, 2004.

**ADDRESSES:** The service information referenced in this AD may be obtained from Construcciones Aeronauticas, S.A., Getafe, Madrid, Spain. This information may be examined at the Federal Aviation Administration (FAA), Transport Airplane Directorate, Rules Docket, 1601 Lind Avenue, SW., Renton, Washington; or at the Office of the Federal Register, 800 North Capitol Street, NW., suite 700, Washington, DC.

**FOR FURTHER INFORMATION CONTACT:** Dan Rodina, Aerospace Engineer, International Branch, ANM-116, FAA, Transport Airplane Directorate, 1601 Lind Avenue, SW., Renton, Washington 98055-4056; telephone (425) 227-2125; fax (425) 227-1149.

**SUPPLEMENTARY INFORMATION:** A proposal to amend part 39 of the Federal Aviation Regulations (14 CFR part 39) to include an airworthiness directive (AD) that is applicable to certain CASA Model C-235 series airplanes was published in the Federal Register on February 6, 2004 (69 FR 5780). That action proposed to require modification of the electrical wiring of the rudder trim control unit.

### **Comments**

We provided the public the opportunity to participate in the development of this AD. No comments have been submitted on the proposed AD or on the determination of the cost to the public.

### **Conclusion**

We have carefully reviewed the available data and determined that air safety and the public interest require adopting the AD as proposed.

# **Cost Impact**

The FAA estimates that 1 airplane of U.S. registry will be affected by this AD, that it will take approximately 7 work hours per airplane to accomplish the required actions, and that the average labor rate is \$65 per work hour. Required parts will cost approximately \$40 per airplane. Based on these figures, the cost impact of the AD on U.S. operators is estimated to be \$495.

The cost impact figure discussed above is based on assumptions that no operator has yet accomplished any of the requirements of this AD action, and that no operator would accomplish those actions in the future if this AD were not adopted. The cost impact figures discussed in AD rulemaking actions represent only the time necessary to perform the specific actions actually required by the AD. These figures typically do not include incidental costs, such as the time required to gain access and close up, planning time, or time necessitated by other administrative actions.

### **Regulatory Impact**

The regulations adopted herein will not have a substantial direct effect on the States, on the relationship between the national Government and the States, or on the distribution of power and responsibilities among the various levels of government. Therefore, it is determined that this final rule does not have federalism implications under Executive Order 13132.

For the reasons discussed above, I certify that this action (1) is not a "significant regulatory action" under Executive Order 12866; (2) is not a "significant rule" under DOT Regulatory Policies and Procedures (44 FR 11034, February 26, 1979); and (3) will not have a significant economic impact, positive or negative, on a substantial number of small entities under the criteria of the Regulatory Flexibility Act. A final evaluation has been prepared for this action and it is contained in the Rules Docket. A copy of it may be obtained from the Rules Docket at the location provided under the caption ADDRESSES.

### List of Subjects in 14 CFR Part 39

Air transportation, Aircraft, Aviation safety, Incorporation by reference, Safety.

### **Adoption of the Amendment**

Accordingly, pursuant to the authority delegated to me by the Administrator, the Federal Aviation Administration amends part 39 of the Federal Aviation Regulations (14 CFR part 39) as follows:

# PART 39-AIRWORTHINESS DIRECTIVES

1. The authority citation for part 39 continues to read as follows:

Authority: 49 U.S.C. 106(g), 40113, 44701.

# § 39.13 [Amended]

2. Section 39.13 is amended by adding the following new airworthiness directive:

# AIRWORTHINESS DIRECTIVE



Aircraft Certification Service Washington, DC

U.S. Department of Transportation Federal Aviation Administration

#### We post ADs on the internet at "www.faa.gov"

The following Airworthiness Directive issued by the Federal Aviation Administration in accordance with the provisions of Title 14 of the Code of Federal Regulations (14 CFR) part 39, applies to an aircraft model of which our records indicate you may be the registered owner. Airworthiness Directives affect aviation safety and are regulations which require immediate attention. You are cautioned that no person may operate an aircraft to which an Airworthiness Directive applies, except in accordance with the requirements of the Airworthiness Directive (reference 14 CFR part 39, subpart 39.3).

### **COMMENT:**

- 1. There is a typo in the AD number in today's Federal Register, pg 17905, 3rd column of this AD. The correct AD number is 2004-07-16. *We have corrected this copy and the pdf copy.*
- 2. There is also a typo in the "applicability" paragraph of this AD. The correct model should read "CN-235". We *aren't correcting* the model in the AD at this time. We are considering the appropriate future action to this AD and will issue the action in the Federal Register at a later date.

**2004-07-16 Construcciones Aeronauticas, S.A. (CASA):** Amendment 39-13560. Docket 2002-NM-160-AD.

**Applicability:** Model C-235 series airplanes, serial numbers C-006, C-007, C-010, C-012, C-018, C-029, C-030, C-032, C-033, and C-042; certificated in any category.

Compliance: Required as indicated, unless accomplished previously.

To prevent the flight crew from being able to inhibit the aural warning for the landing gear up, and the possibility that the flight crew of the next flight or possibly of the same flight could inadvertently land the airplane with the landing gear not down and locked; accomplish the following:

### **Modification**

(a) Within 6 months after the effective date of this AD, modify the electrical wiring of the rudder trim control unit per the Accomplishment Instructions of CASA Service Bulletin SB-235-27-20, dated March 7, 2001.

### **Alternative Methods of Compliance**

(b) In accordance with 14 CFR 39.19, the Manager, International Branch, FAA, Transport Airplane Directorate, is authorized to approve alternative methods of compliance for this AD.

### **Alternative Methods of Compliance**

(c) In accordance with 14 CFR 39.19, the Manager, International Branch, FAA, Transport Airplane Directorate, is authorized to approve alternative methods of compliance for this AD.

# **Incorporation by Reference**

(d) The actions must be done in accordance with CASA Service Bulletin SB-235-27-20, dated March 7, 2001. This incorporation by reference was approved by the Director of the Federal Register in accordance with 5 U.S.C. 552(a) and 1 CFR part 51. Copies may be obtained from Construcciones Aeronauticas, S.A., Getafe, Madrid, Spain. Copies may be inspected at the FAA, Transport Airplane Directorate, 1601 Lind Avenue, SW., Renton, Washington; or at the Office of the Federal Register, 800 North Capitol Street, NW., suite 700, Washington, DC.

**Note 1:** The subject of this AD is addressed in Spanish airworthiness directive 02/02, dated April 30, 2002.

### **Effective Date**

(e) This amendment becomes effective on May 11, 2004.

Issued in Renton, Washington, on March 25, 2004.

Kalene C. Yanamura,

Acting Manager, Transport Airplane Directorate, Aircraft Certification Service.

[FR Doc. 04-7353 Filed 4-5-04; 8:45 am]

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